Architectural Structures I: Statics and Strength of Materials ends 231 Dr. Anne Nichols Spring 2008

Seventeen

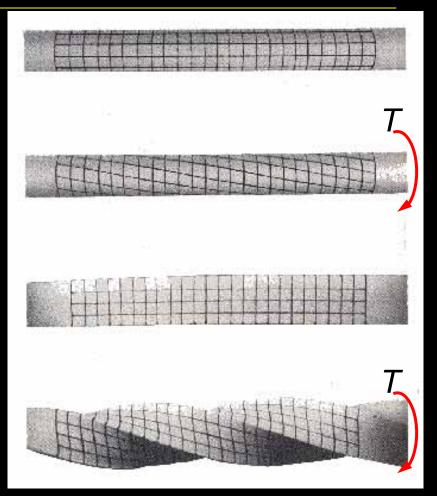
torsion & thermal effects

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Torsional Stress & Strain

- can see torsional stresses & twisting of axi-symmetrical cross sections
 - torque
 - remain plane
 - undistorted
 - rotates
- not true for square sections....



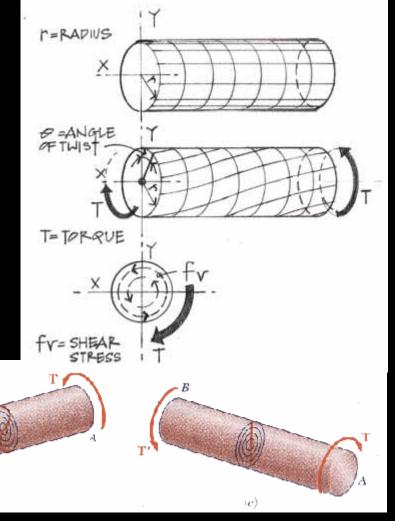
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Shear Stress Distribution

- depend on the deformation
- $\phi = angle of twist$ - measure

a

 can prove planar section doesn't distort

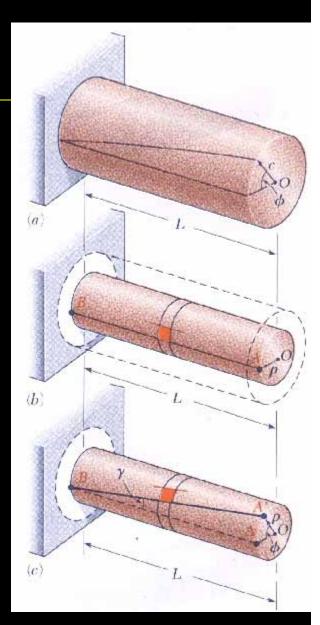


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(b)

Shearing Strain

- related to ϕ
- *ρ* is the radial distance from the centroid to the point under strain
- shear strain varies linearly along the radius: γ_{max} is at outer diameter



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Torsional Stress - Strain

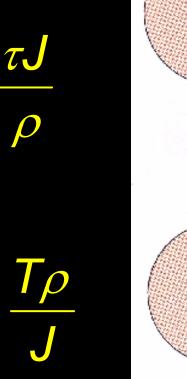
• know
$$f_{\nu} = \tau = G \cdot \gamma$$
 and $\gamma = \frac{\rho \phi}{L}$
• so $\tau = G \cdot \frac{\rho \phi}{L}$

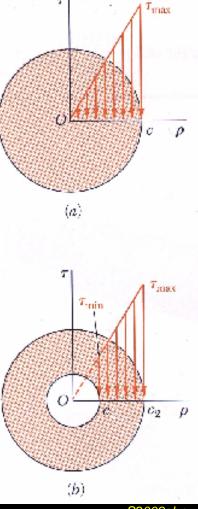
• where G is the <u>Shear Modulus</u>

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Torsional Stress - Strain

- from $T = \Sigma \tau(\rho) \Delta A$ • can derive $T = \frac{\tau J}{\rho}$
 - where J is the polar moment of inertia
 elastic range



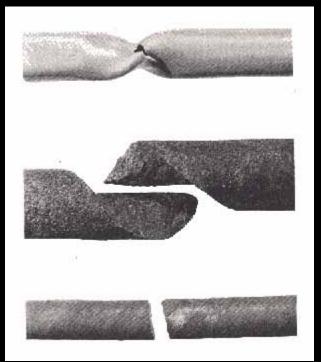


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Shear Stress

- τ_{max} happens at <u>outer diameter</u>
- combined shear and axial stresses

 maximum shear stress at 45° "twisted" plane



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Shear strain

• knowing $\tau = G \cdot \frac{\rho \phi}{L}$ and $\tau = \frac{T\rho}{J}$ • solve: $\phi = \frac{TL}{JG}$

• composite shafts: $\phi = \Sigma$

 $\phi = \sum_{i} \frac{T_{i}L_{i}}{J_{i}G_{i}}$

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Noncircular Shapes

- torsion depends on J
- plane sections don't remain plane
- τ_{max} is still at outer diameter

$$\tau_{\max} = \frac{T}{c_1 a b^2} \quad \phi = \frac{TL}{c_2 a b^3 G}$$

where a is longer side (> b)

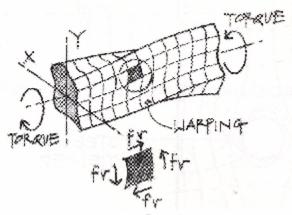


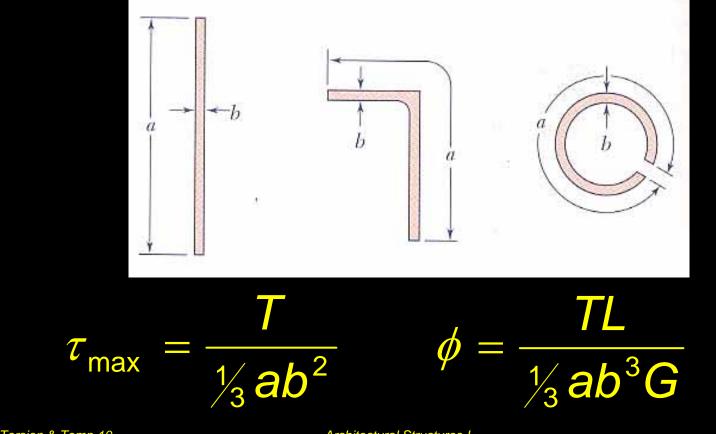
TABLE 3.1. Coefficients for Rectangular Bars in Torsion

a/b	<i>c</i> ₁	<i>C</i> ₂
1.0	° 0.208	0.1406
1.2	0.219	0.1661
1.5	0.231	0.1958
2.0	0.246	0.229
2.5	0.258	0.249
3.0	0.267	0.263
4.0	0.282	0.281
5.0	0.291	0.291
10.0	0.312	0.312
∞	0.333	0.333

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Open Thin-Walled Sections

• with very large a/b ratios:

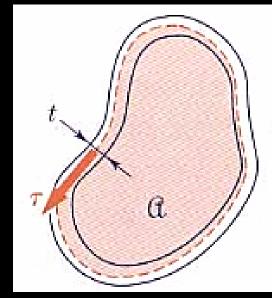


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Shear Flow in Closed Sections

• q is the internal shear force/unit length

 $\tau = \frac{T}{2ta}$ $\phi = \frac{TL}{4ta^2} \sum_{i} \frac{s_i}{t_i}$

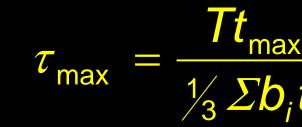


a is the area bounded by the centerline *s_i* is the length segment, *t_i* is the thickness

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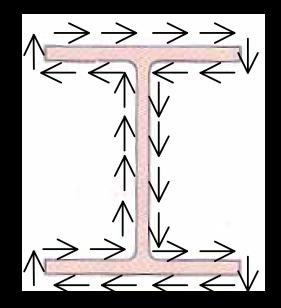
Shear Flow in Open Sections

 each segment has proportion of T with respect to torsional rigidity,



• total angle of twist:

$$\phi = \frac{TL}{\frac{1}{3}G\Sigma b_i t_i^3}$$



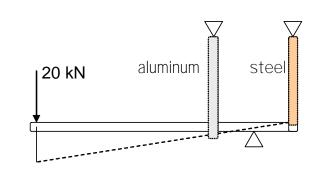
• I beams - web is thicker, so τ_{max} is in web

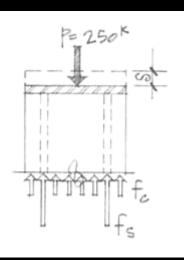
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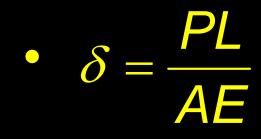
Deformation Relationships

physical movement

 axially (same or zero)
 rotations from axial changes







relates δ to P

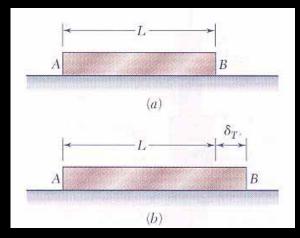
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Deformations from Temperature

- atomic chemistry reacts to changes in energy
- solid materials



- can contract with decrease in temperature
- can expand with increase in temperature
- linear change can be measured per degree



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Thermal Deformation

- α the rate of strain per degree
- UNITS : / , / °F , / °C
- length change:

$$\delta_T = \alpha (\Delta T) L$$

• thermal strain:

$$\varepsilon_T = \alpha(\Delta T)$$

- no stress when movement allowed

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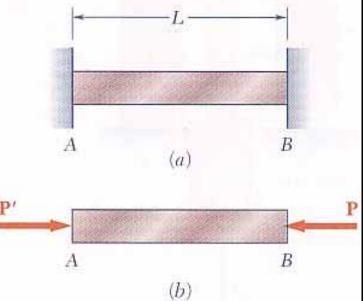
Coefficients of Thermal Expansion

Material	Coefficients (α) [in.	/in./°F]
Wood	3.0 x 10 ⁻⁶	
Glass	4.4 x 10 ⁻⁶	BEARING WALL
Concrete	5.5 x 10 ⁻⁶	JOINT
Cast Iron	5.9 x 10 ⁻⁶	North and a state of the state
Steel	6.5 x 10 ⁻⁶	40 +
Wrought Iron	6.7 x 10 ⁻⁶	40
Copper	<i>9.3 x 10</i> -6	40'
Bronze	10.1 x 10 ⁻⁶	
Brass	10.4 x 10 ⁻⁶	
Aluminum Torsion & Temp 16 Lecture 17	12.8 x 10⁻⁶ Architectural Structures I ENDS 231	S2008abn

Stresses and Thermal Strains

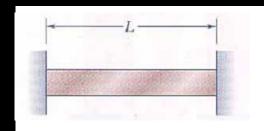
- if thermal movement is restrained
 <u>stresses</u> are induced
- 1. bar pushes on supports
- 2. support pushes back
- 3. reaction causes internal stress $f = \frac{P}{A} = \frac{\delta}{L}E$

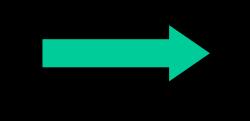
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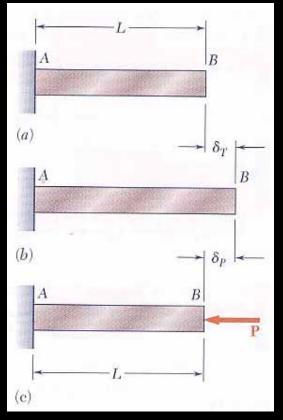


Superposition Method

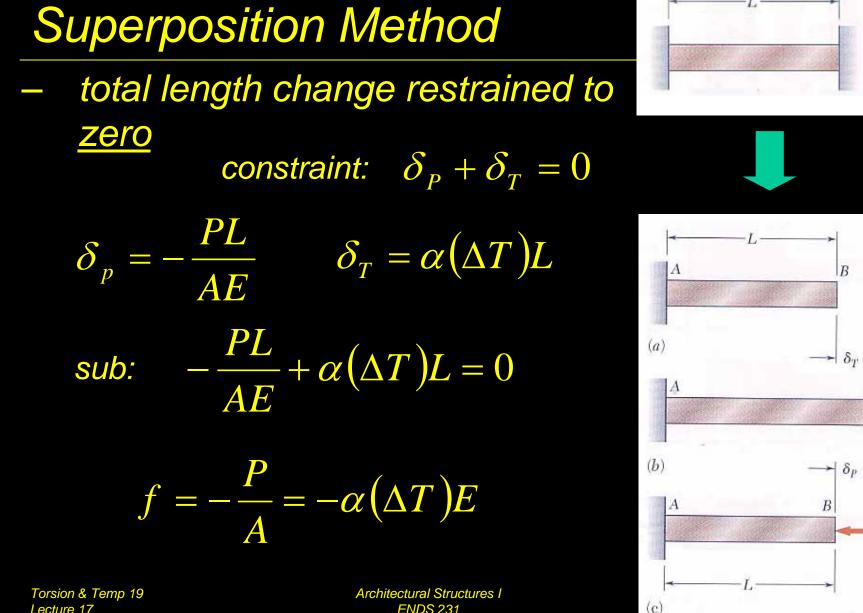
- can remove a support to make it look determinant
- replace the support with a reaction
- enforce the geometry constraint







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B

P

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