



# mechanics of materials

## Mechanics of Materials

- MECHANICS

- MATERIALS



## Mechanics of Materials

- external loads and their effect on deformable bodies
- use it to answer question if structure meets requirements of
  - stability and equilibrium
  - strength and stiffness
- other principle building requirements
  - economy, functionality and aesthetics

## Knowledge Required

- material properties
- member cross sections
- ability of a material to resist breaking
- structural elements that resist excessive
  - deflection
  - deformation

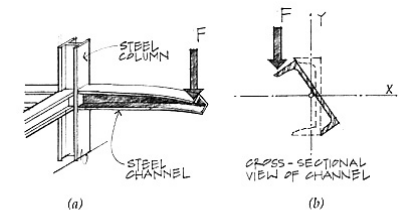


Figure 2.34 An example of torsion on a cantilever beam.

## Problem Solving

### 1. STATICS:

equilibrium of external forces,  
internal forces, stresses

### 2. GEOMETRY:

cross section properties, deformations and  
conditions of geometric fit, strains

### 3. MATERIAL PROPERTIES:

stress-strain relationship for each material  
obtained from testing



## Design

- materials have a critical stress value  
where they could break or yield

- ultimate stress
- yield stress
- compressive stress
- fatigue strength
- (creep & temperature)

acceptance  
vs. failure

## Stress

- stress is a term for the intensity of a  
force, like a pressure
- internal or applied
- force per unit area

$$\text{stress} = f = \frac{P}{A}$$



## Design (cont)

- we'd like

$$f_{\text{actual}} \ll F_{\text{allowable}}$$

- stress distribution may  
vary: average
- uniform distribution  
exists IF the member is  
loaded axially  
(concentric)

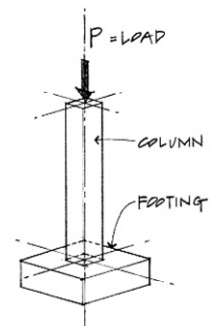
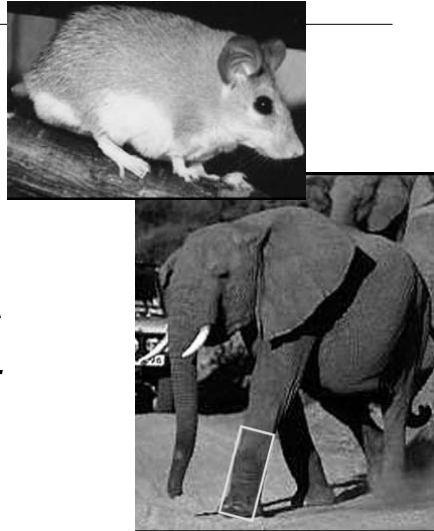


Figure 5.3 Centric loads.

## Scale Effect

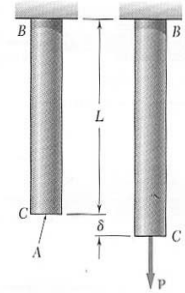
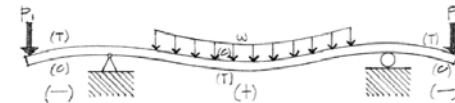
- *model scale*
  - material weights, small areas
- *structural scale*
  - much more material weight, bigger areas
- *ratio is not constant:*

$$\frac{\gamma L^3}{L^2} = \gamma L$$



## Strain

- *materials deform*
- *axially loaded materials change length*
- *bending materials deflect*



- **STRAIN:**
  - change in length over length

$$\text{strain} = \varepsilon = \frac{\Delta L}{L}$$

## Normal Stress

- normal stress is normal to the cross section
  - stressed area is perpendicular to the load

$$f_{t \text{ or } c} = \frac{P}{A}$$

( $\sigma$ )

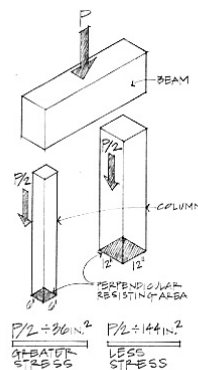


Figure 5.7 Two columns with the same load, different stress.

## Shear Stress

- *stress parallel to a surface*

$$f_v = \frac{P}{A} = \frac{P}{td}$$

( $\tau_{ave}$ )

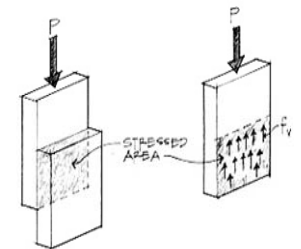


Figure 5.10 Shear stress between two glued blocks.

## Bearing Stress

- stress on a surface by contact in compression

$$f_p = \frac{P}{A} = \frac{P}{td}$$

( $\sigma$ )

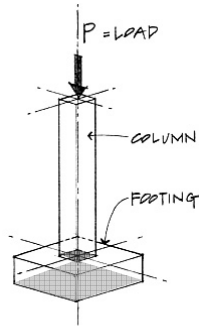


Figure 5.3 Centric loads.

## Bending Stress

- normal stress caused by bending

$$f_b = \frac{Mc}{I} = \frac{M}{S}$$

( $\sigma$ )

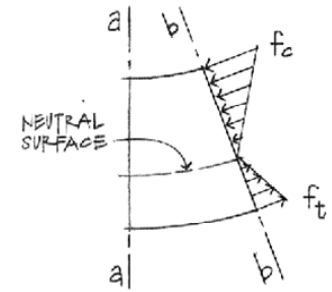


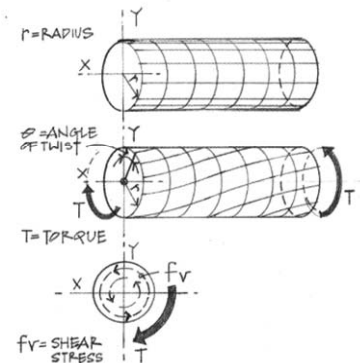
Figure 8.8 Bending stresses on section b-b.

## Torsional Stress

- shear stress caused by twisting

$$f_v = \frac{T\rho}{J}$$

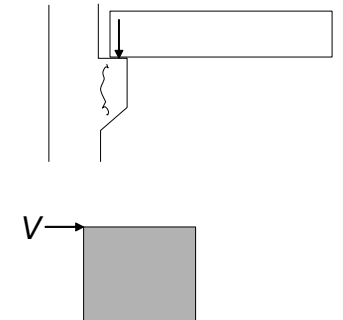
( $\tau$ )



## Structures and Shear

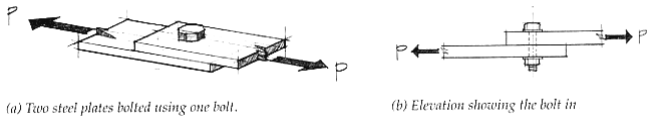
- what structural elements see shear?

- beams
  - bolts
  - splices
  - slabs
  - footings
  - walls
  - wind
  - seismic loads
- } connections



# Bolts

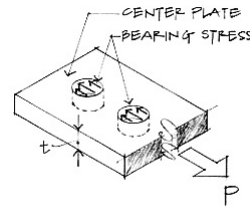
- connected members in tension cause shear stress



(a) Two steel plates bolted using one bolt.

(b) Elevation showing the bolt in shear.

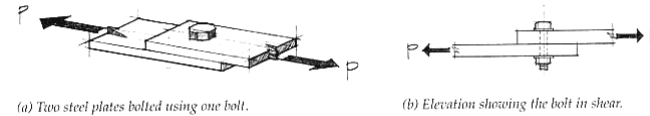
- connected members in compression cause bearing stress



Bearing stress on plate.

# Single Shear

- seen when 2 members are connected



(a) Two steel plates bolted using one bolt.

(b) Elevation showing the bolt in shear.

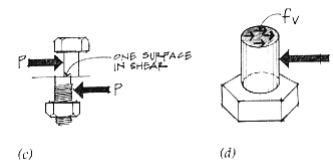


Figure 5.11 A bolted connection—single shear.

$f_v$  = Average shear stress through bolt cross section

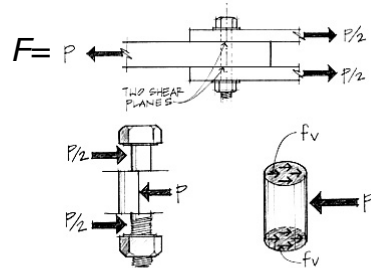
$A$  = Bolt cross-sectional area

$$f_v = \frac{P}{A}$$

$$f_v = \frac{P}{A} = \frac{P}{\pi d^2/4}$$

# Double Shear

- seen when 3 members are connected
- two areas



Free-body diagram of middle section of the bolt in shear.

Figure 5.12 A bolted connection in double shear.

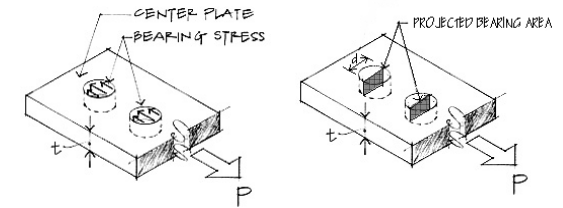
$$f_v = \frac{P}{2A}$$

(two shear planes)

$$f_v = \frac{P}{2A} = \frac{P/2}{A} = \frac{P/2}{\pi d^2/4}$$

# Bolt Bearing Stress

- compression & contact
- projected area



Bearing stress on plate.

$$f_p = \frac{P}{A_{projected}} = \frac{P}{td}$$