

# torsion & thermal effects

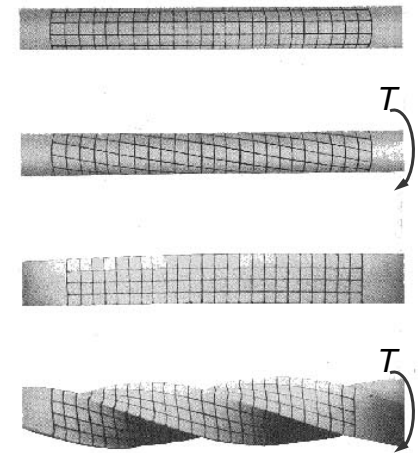
Torsion & Temp 1  
Lecture 17

Architectural Structures I  
ENDS 231

F2005abn

## Torsional Stress & Strain

- can see torsional stresses & twisting of axi-symmetrical cross sections
  - torque
  - remain plane
  - undistorted
  - rotates
- not true for square sections....



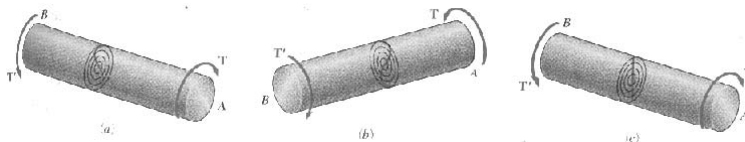
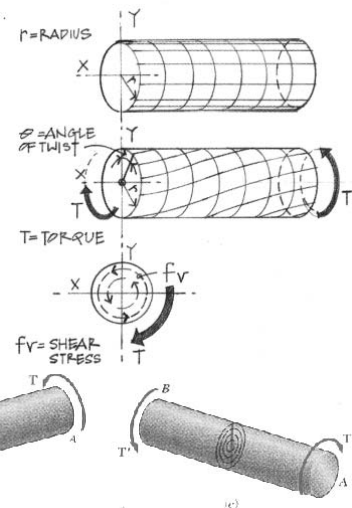
Torsion & Temp 7  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Shear Stress Distribution

- depend on the deformation
- $\phi$  = angle of twist
  - measure
- can prove planar section doesn't distort



Torsion & Temp 8  
Lecture 17

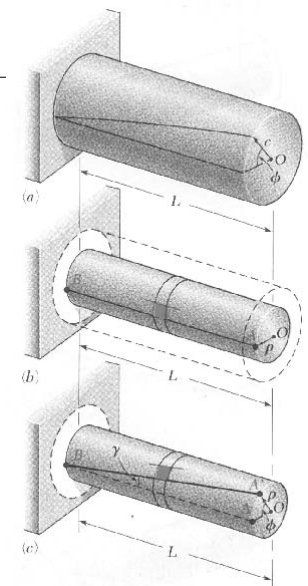
Architectural Structures I  
ENDS 231

S2004abn

## Shearing Strain

- related to  $\phi$ 

$$\gamma = \frac{\rho\phi}{L}$$
- $\rho$  is the radial distance from the centroid to the point under strain
- shear strain varies linearly along the radius:  $\gamma_{max}$  is at outer diameter



Torsion & Temp 9  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Torsional Stress - Strain

- know  $f_v = \tau = G \cdot \gamma$  and  $\gamma = \frac{\rho\phi}{L}$
- so  $\tau = G \cdot \frac{\rho\phi}{L}$
- where  $G$  is the Shear Modulus

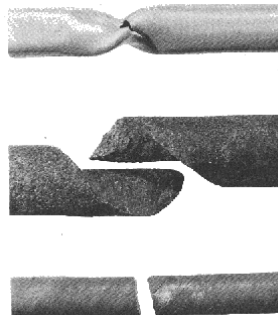
Torsion & Temp 10  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Shear Stress

- $\tau_{max}$  happens at outer diameter
- combined shear and axial stresses
  - maximum shear stress at 45° “twisted” plane



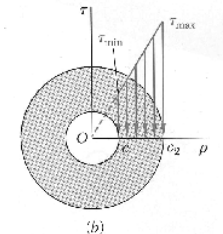
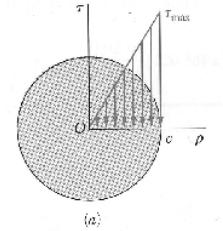
Torsion & Temp 12  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Torsional Stress - Strain

- from  $T = \Sigma \tau(\rho) \Delta A$
- can derive  $T = \frac{\tau J}{\rho}$ 
  - where  $J$  is the polar moment of inertia
  - elastic range  $\tau = \frac{T\rho}{J}$



Torsion & Temp 11  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Shear strain

- knowing  $\tau = G \cdot \frac{\rho\phi}{L}$  and  $\tau = \frac{T\rho}{J}$
- solve:  $\phi = \frac{TL}{JG}$
- composite shafts:  $\phi = \Sigma_i \frac{T_i L_i}{J_i G_i}$

Torsion & Temp 13  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Noncircular Shapes

- torsion depends on  $J$
- plane sections don't remain plane
- $\tau_{max}$  is still at outer diameter

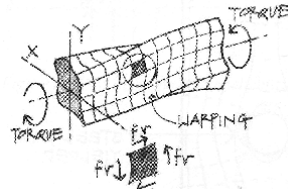


TABLE 3.1. Coefficients for Rectangular Bars in Torsion

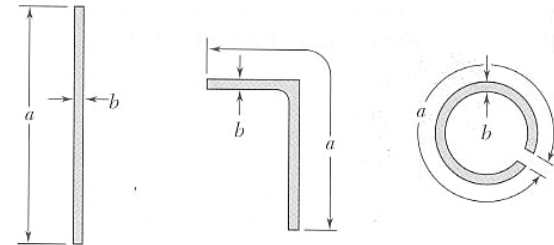
$a/b$	$c_1$	$c_2$
1.0	0.208	0.1406
1.2	0.219	0.1661
1.5	0.231	0.1958
2.0	0.246	0.229
2.5	0.258	0.249
3.0	0.267	0.263
4.0	0.282	0.281
5.0	0.291	0.291
10.0	0.312	0.312
$\infty$	0.333	0.333

$$\tau_{max} = \frac{T}{c_1 ab^2} \quad \phi = \frac{TL}{c_2 ab^3 G}$$

– where  $a$  is longer side ( $> b$ )

## Open Thin-Walled Sections

- with very large  $a/b$  ratios:



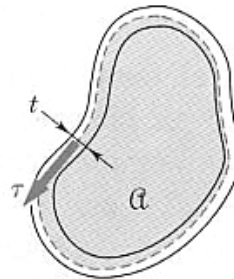
$$\tau_{max} = \frac{T}{\frac{1}{3} ab^2} \quad \phi = \frac{TL}{\frac{1}{3} ab^3 G}$$

## Shear Flow in Closed Sections

- $q$  is the internal shear force/unit length

$$\tau = \frac{T}{2t\mathcal{A}}$$

$$\phi = \frac{TL}{4t\mathcal{A}^2} \sum_i \frac{s_i}{t_i}$$



- $\mathcal{A}$  is the area bounded by the centerline
- $s_i$  is the length segment,  $t_i$  is the thickness

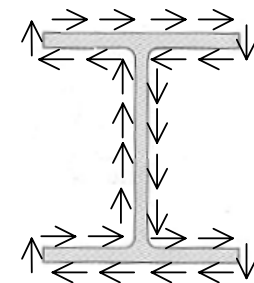
## Shear Flow in Open Sections

- each segment has proportion of  $T$  with respect to torsional rigidity,

$$\tau_{max} = \frac{Tt_{max}}{\frac{1}{3} \sum b_i t_i^3}$$

- total angle of twist:

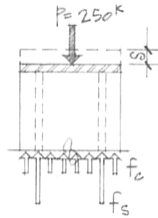
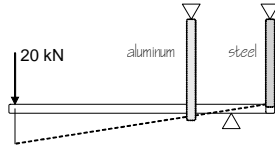
$$\phi = \frac{TL}{\frac{1}{3} G \sum b_i t_i^3}$$



- I beams - web is thicker, so  $\tau_{max}$  is in web

## Deformation Relationships

- physical movement
  - axially (same or zero)
  - rotations from axial changes



- $\delta = \frac{PL}{AE}$  relates  $\delta$  to  $P$

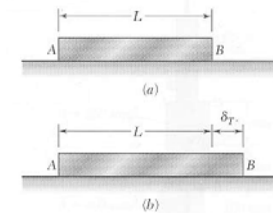
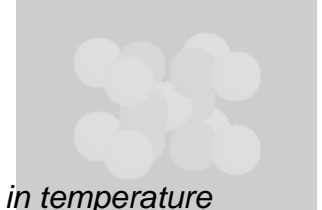
Torsion & Temp 18  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Deformations from Temperature

- atomic chemistry reacts to changes in energy
- solid materials
  - can contract with decrease in temperature
  - can expand with increase in temperature
- linear change can be measured per degree



Torsion & Temp 19  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Thermal Deformation

- $\alpha$  - the rate of strain per degree
- UNITS :  $/^{\circ}F$  ,  $/^{\circ}C$
- length change:  $\delta_T = \alpha(\Delta T)L$
- thermal strain:  $\epsilon_T = \alpha(\Delta T)$ 
  - no stress when movement allowed

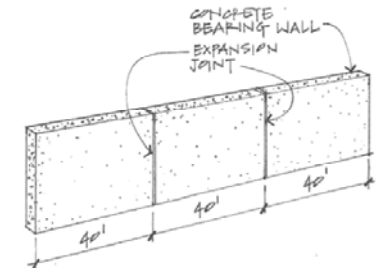
Torsion & Temp 20  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Coefficients of Thermal Expansion

Material	Coefficients ( $\alpha$ ) [in./in./ $^{\circ}F$ ]
Wood	$3.0 \times 10^{-6}$
Glass	$4.4 \times 10^{-6}$
Concrete	$5.5 \times 10^{-6}$
Cast Iron	$5.9 \times 10^{-6}$
Steel	$6.5 \times 10^{-6}$
Wrought Iron	$6.7 \times 10^{-6}$
Copper	$9.3 \times 10^{-6}$
Bronze	$10.1 \times 10^{-6}$
Brass	$10.4 \times 10^{-6}$
Aluminum	$12.8 \times 10^{-6}$



Torsion & Temp 21  
Lecture 17

Architectural Structures I  
ENDS 231

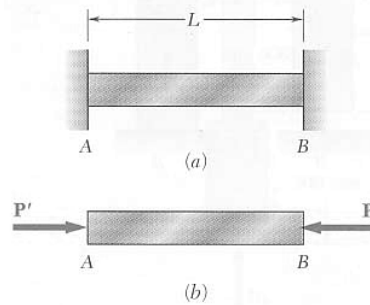
S2004abn

## Stresses and Thermal Strains

- if thermal movement is restrained stresses are induced

- bar pushes on supports
- support pushes back
- reaction causes internal stress

$$f = \frac{P}{A} = \frac{\delta}{L} E$$



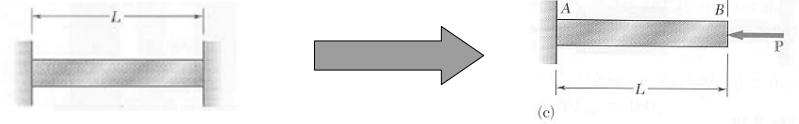
Torsion & Temp 22  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Superposition Method

- can remove a support to make it look determinant
- replace the support with a reaction
- enforce the geometry constraint



Torsion & Temp 23  
Lecture 17

Architectural Structures I  
ENDS 231

S2004abn

## Superposition Method

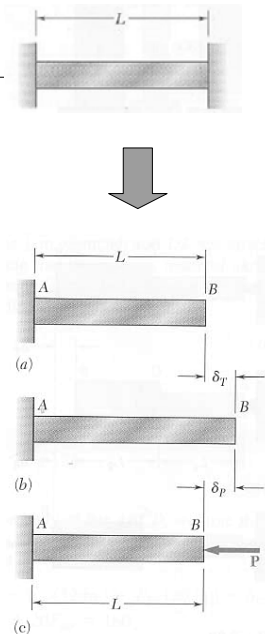
- total length change restrained to zero

$$\text{constraint: } \delta_P + \delta_T = 0$$

$$\delta_P = -\frac{PL}{AE} \quad \delta_T = \alpha(\Delta T)L$$

$$\text{sub: } -\frac{PL}{AE} + \alpha(\Delta T)L = 0$$

$$f = -\frac{P}{A} = -\alpha(\Delta T)E$$



Torsion & Temp 24  
Lecture 17

Architectural Structures I  
ENDS 231