# Wood Design

# Notation:

Nota	tio	n:
а	=	name for width dimension
A	=	name for area
A <sub>reg'd</sub>	-adj	= area required at allowable stress
1	0	when shear is adjusted to include
		self weight
b	=	width of a rectangle
	=	name for height dimension
$c_1$	=	coefficient for shear stress for a
		rectangular bar in torsion
$C_C$	=	curvature factor for laminated
		arches
$C_D$	=	load duration factor
$C_{fu}$	=	flat use factor for other than decks
$C_F$	=	size factor
$C_H$	=	shear stress factor
$C_i$	=	incising factor
$C_L$	=	beam stability factor
$C_M$	=	wet service factor
$C_p$	=	column stability factor for wood
-		design
$C_r$	=	repetitive member factor for wood
		design
$C_V$	=	volume factor for glue laminated
		timber design
$C_t$	=	temperature factor for wood design
d	=	name for depth
$d_{min}$	=	dimension of timber critical for
		buckling
DL	=	shorthand for dead load
E	=	modulus of elasticity
f		stress (strength is a stress limit)
$f_b$		bending stress
$f_{from ta}$		= tabular strength (from table)
$f_p$		bearing stress
$f_r$		radial stress for a glulam timber
$f_v$		shear stress
		maximum shear stress
$F_b$		tabular bending strength
		allowable bending stress
$F_{b}^{\prime}$	=	allowable bending stress (adjusted)
$F_{c}$	=	tabular compression strength
-		parallel to the grain
F <b>'</b>	=	allowable compressive stress
C		(adjusted)
		(aujusicu)

$F^*c$	= intermediate compressive stress for
	column design dependent on load
	duration
$F_{cE}$	= theoretical allowed buckling stress
$F_{c\perp}$	= tabular compression strength
	perpendicular to the grain
$F_p$	= tabular bearing strength parallel to
1	the grain
	= allowable bearing stress
$F_R$	= allowable radial stress
$F_t$	= tabular tensile strength
$F_u$	= ultimate strength
$F_v$	= tabular bending strength
	= allowable shear stress
h	= height of a rectangle
Ι	= moment of inertia with respect to
	neutral axis bending
I <sub>trial</sub>	= moment of inertia of trial section
I <sub>req'd</sub>	= moment of inertia required at
	limiting deflection
$I_y$	= moment of inertia with respect to an
	y-axis
J	= polar moment of inertia
$K_{cE}$	= material factor for wood column
	design
$L_e$	= effective length that can buckle for
	column design, as is $\ell_e$
L	= name for length or span length
LL	= shorthand for live load
LRFI	D = load and resistance factor design
М	= internal bending moment
	= maximum internal bending moment
$M_{max}$	adj = maximum bending moment
	adjusted to include self weight
Р	= name for axial force vector
R	= radius of curvature of a deformed
	beam
	= radius of curvature of a laminated
	arch
G	= name for a reaction force
S	= section modulus
$S_{req'd}$	= section modulus required at
	allowable stress

$S_{req'd-adj}$ = section modulus required at	$w_{selfwt}$ = name for distributed load from self				
allowable stress when moment is	weight of member				
adjusted to include self weight	$\Delta_{allowable}$ = allowable beam deflection				
T = torque (axial moment)	$\Delta_{limit}$ = allowable beam deflection limit				
V = internal shear force	$\Delta_{max}$ = maximum beam deflection				
$V_{max}$ = maximum internal shear force	$\kappa$ = slenderness ratio limit for long				
$V_{max-adj}$ = maximum internal shear force	columns				
adjusted to include self weight	$\gamma$ = density or unit weight				
w = name for distributed load	$\rho$ = radial distance				

#### Wood or Timber Design

Structural design standards for wood are established by the *National Design Specification (NDS)* published by the National Forest Products Association. There is a combined specification (from 2005) for **Allowable** Stress Design and limit state design (LRFD).

Tabulated wood strength values are used as the base allowable strength (ASD) and modified by appropriate adjustment factors:  $f = C_D C_M C_F ... \times f_{from table}$ 

#### Adjustment Factors

CD	load duration factor
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- $C_M$  wet service factor (1.0 dry < 16% moisture content)
- Ct temperature factor (at high temperatures strength decreases)
- C<sub>L</sub> beam stability factor (for beams without full lateral support)
- $C_F$  size factor for visually graded sawn lumber and round timber > 12" depth

$$C_F = (12/d)^{\frac{1}{9}} \le 1.0$$

- FACTOR FERMANENT-DL (Co-0.9) 1.9 DURATION SNOW *|.*8 FLOOR LL 1.7 1.6 6-16 45 14 13 LOAD = 1.25 1.2  $C_{n} = -1.15$ į., C. -1.0 1.0 0.9 ٤ ž 550 NWN ] Ŕ 1 18 QAY 148 ð ŝ DURATION OF LOAD (TIME)
- $C_V$  volume factor for glued laminated timber (similar to  $C_F$ )
- C<sub>fu</sub> flat use factor (excluding decking)
- C<sub>r</sub> repetitive member factor (1.15 for three or more parallel members of Dimension lumber spaced not more than 24 in. on center, connected together by a loaddistributing element such as roof, floor, or wall sheathing)
- C<sub>c</sub> curvature factor for glued laminated timber (1.0 straight & cambered)  $t/R \le 1/100$  for hardwoods & southern pine or 1/125 other softwoods

 $C_{a} = 1 - 2000(t/R)^{2}$ 

- C<sub>i</sub> incising factor (0.85 incised sawn lumber, 1 for sawn lumber not incised and glulam)
- C<sub>H</sub> shear stress factor (amount of splitting)
- C<sub>P</sub> column stability factor (1.0 for fully supported columns)

#### Design Values

- $F_{\rm b}$ : bending stress
- F<sub>t</sub>: tensile stress
- F<sub>v</sub>: horizontal shear stress
- $F_{c\perp}$ : compression stress (perpendicular to grain)
- F<sub>c</sub>: compression stress (parallel to grain)
- E: modulus of elasticity
- F<sub>p</sub>: bearing stress (parallel to grain)

Wood is significantly weakest in shear and strongest along the direction of the grain (tension and compression).

# Load Combinations and Deflection

The critical load combination is determined by the largest of either:

 $\frac{dead \ load}{0.9} \ or \frac{(\ dead \ load + any \ combination \ of \ live \ load \ )}{C_D}$ 

The deflection limits may be increased for less stiffness with total load: LL + 0.5(DL)

# Criteria for Beam Design

Allowable normal stress or normal stress from LRFD should not be exceeded:

Knowing M and F<sub>b</sub>, the minimum section modulus fitting the limit is:

$$S_{req'd} \ge \frac{M}{F_h}$$

Besides strength, we also need to be concerned about *serviceability*. This involves things like limiting deflections & cracking, controlling noise and vibrations, preventing excessive settlements of foundations and durability. When we know about a beam section and its material, we can determine beam deformations.

# Determining Maximum Bending Moment

Drawing V and M diagrams will show us the maximum values for design. Computer applications are very helpful.

# Determining Maximum Bending Stress

For a prismatic member (constant cross section), the maximum normal stress will occur at the maximum moment.

For a *non-prismatic* member, the stress varies with the cross section AND the moment.

#### Deflections

Elastic curve equations can be found in handbooks, textbooks, design manuals, etc...Computer programs can be used as well.

Elastic curve equations can be superpositioned ONLY if the stresses are in the elastic range. The deflected shape is roughly the same shape flipped as the bending moment diagram but is constrained by supports and geometry.

#### Allowable Deflection Limits

All building codes and design codes limit deflection for beam types and damage that could happen based on service condition and severity.

Use	LL only	DL+LL	
Roof beams:			
Industrial	L/180	L/120	
Commercial			
plaster ceiling	L/240	L/180	
no plaster	L/360	L/240	
Floor beams:			
Ordinary Usage	L/360	L/240	
Roof or floor (damageable	e elements)	L/480	

#### Lateral Buckling

With compression stresses in the top of a beam, a sudden "popping" or buckling can happen even at low stresses. In order to prevent it, we need to brace it along the top, or laterally brace it, or provide a bigger  $I_y$ .

#### Beam Loads & Load Tracing

In order to determine the loads on a beam (or girder, joist, column, frame, foundation...) we can start at the top of a structure and determine the *tributary area* that a load acts over and the beam needs to support. Loads come from material weights, people, and the environment. This area is assumed to be from half the distance to the next beam over to halfway to the next beam.

The reactions must be supported by the next lower structural element *ad infinitum*, to the ground.

#### Design Procedure

The intent is to find the most light weight member satisfying the section modulus size.

- 1. Know  $F_{all}$  for the material or  $F_U$  for LRFD.
- 2. Draw V & M, finding  $M_{max}$ .

**ARCH 631** 

- 3. Calculate S<sub>req'd</sub>. This step is equivalent to determining  $f_b = \frac{M_{\text{max}}}{S} \le F'_b$  $S = \frac{bh^2}{c}$
- 4. For rectangular beams
  - For timber: use the section charts to find S that will work and remember that the beam self weight will increase S<sub>reg'd.</sub>

\*\*\*\*Determine the "updated"  $V_{max}$  and  $M_{max}$  including the beam self weight, and verify that the updated S<sub>reg'd</sub> has been met.\*\*\*\*\*

5. Consider lateral stability.

6. Evaluate horizontal shear stresses using  $V_{max}$  to determine if  $f_v \le F'_v$ 

For rectangular beams

$$f_{v-\max} = \frac{3V}{2A} = 1.5 \frac{V}{A}$$

- 7. Provide adequate bearing area at supports:
- 8. Evaluate shear due to torsion

$$f_{\nu} = \frac{T\rho}{J} \text{ or } \frac{T}{c_1 a b^2} \le F_{\nu}'$$

 $f_p = \frac{P}{\Lambda} \le F'_p$ 

(circular section or rectangular)

- 9. Evaluate the deflection to determine if  $\Delta_{maxLL} \leq \Delta_{LL-allowed}$  and/or  $\Delta_{maxTotal} \leq \Delta_{Total-allowed}$
- \*\*\*\* note: when  $\Delta_{calculated} > \Delta_{limit}$ ,  $I_{required}$  can be found with: and S<sub>req'd</sub> will be satisfied for similar self weight \*\*\*\*\*

$$I_{req'd} \ge \frac{\Delta_{too \, big}}{\Delta_{lim \, it}} I_{trial}$$

#### FOR ANY EVALUATION:

Redesign (with a new section) at any point that a stress or serviceability criteria is NOT satisfied and re-evaluate each condition until it is satisfactory.

# **Column Design**

National Design Specification for Wood Construction (1992):

Any slenderness ratio,  $l_e/d \le 50$ :

$$f_c = \frac{P}{A} \le F_c'$$

$$F_c' = F_c(C_D)(C_M)(C_t)(C_F)(C_p)$$

For preliminary column design:

$$F_c' = F_c^* C_p = (F_c C_D) C_p$$

# Procedure

1. Obtain F'<sub>c</sub>

find  $l_e/d$  or assume  $(l_e/d \le 50)$ compute  $F_{cE} = \frac{K_{cE}E}{\binom{l_e}{d}^2}$  with  $K_{cE} = 0.3$  for sawn, = 0.418 for glu-lam compute  $F_c^* \cong F_c C_D$  with  $C_D = 1$ , normal,  $C_D = 1.25$  for 7 day roof... find  $F_{cE}/F_c^*$  and get  $C_p$ 

2. Select a section

If the load and area of the column are known, set the stress equal to the allowable stress, and solve for  $l_{e}$ , l, or  $d_{min}$ 

If the load and length of the column are known, set the stress equal to the allowable stress, and solve for A or  $d_{min}$  and select a section that satisfies the values found.

3. Continue from 2 until  $F'_c$  is satisfied:  $F'_c = F^*_c C_p$ 

# Alternate Column Allowable Stress

For intermediate length columns with 11<L/d< $\kappa$ , where  $\kappa = 0.67 \sqrt{\frac{E}{F_c}}$ :

$$F_{c}' = F_{c} \left\{ 1 - \left( \frac{1}{3} \left[ \left( \frac{L_{e}}{d} \right) \kappa \right]^{4} \right\} \right\}$$

For *long* columns with  $L/d > \kappa$ , and <u>an assumed safety factor of 2.73</u>: the allowable stress is:

$$F_c = \frac{0.3E}{\left(\frac{L_e}{d}\right)^2}$$

# Table 9.3Column stability factor $C_p$ .

Statics and Strength of Materials for Architecture and Building Construction, 2nd ed., Onouye & Kane

# Column Stability Factor Cp

	$\mathbf{P}_{p}^{\mathbf{n}} = \mathbf{C}_{p} \cdot \mathbf{F}_{c}^{\mathbf{n}}$	$F_{CE} = \frac{.30 \text{ E}}{(1/\text{d})^2}$ for sawn posts	$F_{CE} = \frac{418 E}{(1/d)^2}$ for Glu-Lam posts
$\frac{F_{CE}}{F_{C}^{e}} = \begin{array}{c} \text{Sawn} & \text{Glu-Lam} \\ \hline F_{C}^{e} & C_{p} & C_{p} \end{array}$	$ \begin{array}{ c c c }\hline F_{CE} & Sawn & Glu-Lam \\\hline F_{C}^{\bullet} & C_{p} & C_{p} \end{array} $	$ \begin{array}{ c c c }\hline F_{CF} & Sawn & Glu-Lam \\\hline F_{C}^{*} & C_{p} & C_{p} \end{array} $	$ \begin{array}{c c} F_{CE} & Sawn & Glu-Lam \\ \hline F_{C}^{*} & C_{p} & C_{p} \end{array} $
$\begin{array}{ccccccc} 0.00 & 0.000 & 0.000 \\ 0.01 & 0.010 & 0.010 \\ 0.02 & 0.020 & 0.320 \\ 0.03 & 0.030 & 0.030 \\ 0.04 & 0.040 & 0.040 \\ 0.05 & 0.049 & 0.050 \\ 0.06 & 0.059 & 0.060 \\ 0.07 & 0.069 & 0.069 \\ 0.08 & 0.079 & 0.079 \\ 0.09 & 0.088 & 0.089 \\ \end{array}$	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	2.40 0.894 0.940   2.45 0.897 0.941   2.50 0.899 0.943   2.55 0.901 0.944   2.60 0.904 0.946   2.65 0.906 0.947   2.70 0.908 0.949   2.75 0.910 0.950   2.80 0.912 0.951   2.85 0.914 0.952
0.10 0.098 0.099   0.11 0.107 0.109   0.12 0.117 0.118   0.13 0.126 0.128   0.14 0.136 0.138   0.15 0.145 0.147   0.16 0.154 0.157   0.17 0.164 0.167   0.18 0.173 0.176   0.19 0.182 0.186	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	1.40 0.793 0.862   1.42 0.796 0.865   1.44 0.800 0.868   1.46 0.803 0.871   1.48 0.807 0.874   1.50 0.810 0.877   1.52 0.813 0.879   1.54 0.816 0.882   1.56 0.819 0.884   1.58 0.822 0.887	$\begin{array}{cccccccccccccccccccccccccccccccccccc$
$\begin{array}{ccccccc} 0.20 & 0.191 & 0.195 \\ 0.21 & 0.200 & 0.205 \\ 0.22 & 0.209 & 0.214 \\ 0.23 & 0.218 & 0.224 \\ 0.24 & 0.227 & 0.233 \\ 0.25 & 0.235 & 0.242 \\ 0.26 & 0.244 & 0.252 \\ 0.27 & 0.253 & 0.261 \\ 0.28 & 0.261 & 0.270 \\ 0.29 & 0.270 & 0.279 \\ \end{array}$	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	1.60 0.825 0.889   1.62 0.827 0.891   1.64 0.830 0.893   1.66 0.832 0.895   1.68 0.835 0.897   1.70 0.837 0.899   1.72 0.840 0.901   1.74 0.842 0.903   1.76 0.844 0.904   1.78 0.846 0.906	$\begin{array}{cccccccccccccccccccccccccccccccccccc$
$\begin{array}{cccccc} 0.30 & 0.278 & 0.288 \\ 0.31 & 0.287 & 0.297 \\ 0.32 & 0.295 & 0.306 \\ 0.33 & 0.304 & 0.315 \\ 0.34 & 0.312 & 0.324 \\ 0.35 & 0.320 & 0.333 \\ 0.36 & 0.328 & 0.342 \\ 0.37 & 0.336 & 0.351 \\ 0.38 & 0.344 & 0.360 \\ 0.39 & 0.352 & 0.368 \\ \end{array}$	$\begin{array}{ccccccc} 0.90 & 0.653 & 0.718 \\ 0.91 & 0.658 & 0.722 \\ 0.92 & 0.661 & 0.727 \\ 0.93 & 0.665 & 0.731 \\ 0.94 & 0.669 & 0.735 \\ 0.95 & 0.673 & 0.740 \\ 0.96 & 0.677 & 0.744 \\ 0.97 & 0.680 & 0.748 \\ 0.98 & 0.684 & 0.752 \\ 0.99 & 0.688 & 0.756 \\ \end{array}$	1.80 0.849 0.908   1.82 0.851 0.909   1.84 0.853 0.911   1.86 0.855 0.912   1.88 0.857 0.914   1.90 0.858 0.915   1.92 0.860 0.916   1.94 0.862 0.919   1.98 0.864 0.919	$\begin{array}{cccccccccccccccccccccccccccccccccccc$
0.40 0.360 0.377   0.41 0.367 0.386   0.42 0.375 0.394   0.43 0.383 0.403   0.44 0.390 0.411   0.45 0.398 0.420   0.46 0.405 0.428   0.46 0.4012 0.436   0.46 0.402 0.428   0.47 0.412 0.436   0.48 0.419 0.444   0.49 0.427 0.453	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	2.00 0.867 0.921   2.02 0.869 0.922   2.04 0.870 0.924   2.06 0.872 0.925   2.08 0.874 0.926   2.10 0.875 0.927   2.12 0.876 0.929   2.16 0.879 0.930   2.18 0.881 0.931	4.40 0.948 0.972   4.45 0.949 0.973   4.50 0.949 0.973   4.55 0.950 0.974   4.60 0.950 0.974   4.65 0.951 0.974   4.70 0.952 0.975   4.75 0.952 0.975   4.80 0.953 0.975
$\begin{array}{cccccccccccccccccccccccccccccccccccc$		2.20 0.882 0.932   2.22 0.883 0.932   2.24 0.885 0.933   2.26 0.886 0.934   2.28 0.887 0.935   2.30 0.888 0.936   2.32 0.889 0.937   2.34 0.891 0.937   2.36 0.892 0.938   2.38 0.893 0.939	4.90 0.954 0.976   5.00 0.955 0.976   6.00 0.963 0.981   8.00 0.973 0.986   10.00 0.979 0.989   20.00 0.990 0.995   40.00 0.995 0.997   60.00 0.997 0.998   100.00 0.997 0.998   100.00 0.998 0.999   200.00 0.999 0.999

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considerations will permit, the finished sizes of structural glued laminated timber as given in Table B constitute normal industry practice. Industry standards do, however, permit the use of any depth or width of glued laminated timber. Dimension lumber of 1½ in. net thickness is normally used for laminating straight members. The modified section modulus includes size factor (C <sub>F</sub> ), and no further reduction of bending stress for size is needed.											
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		<u>ğ</u> .c		1		SECTION 5, SC <sub>F</sub> in. <sup>3</sup>		1 · · · ·		SECTION SC In 3	
مر	۵.	ក្តស្វ	ŗ.	-	۰.	u u u u u u u u u u u u u u u u u u u	Ъ́с	i e	<b>.</b> (	្អរអ្នក	۳.
d T	AREA, A in. <sup>e</sup>	C S	5	DEPTH, d in.	AREA, A in. <sup>®</sup>	8 S	E Z	Verin, a in.	AREA, A In."	<b>G</b> 2	E Z
Ę	₹	DOIE	STILA	Ŧ	Ś	ED	<b>ME</b>	Ę	₹	LEO	E S S
DEPTH,	ARE	MODIFIED SECTION MODULUS, SCF In.3	MOMENT C INERTIA, I i	DEP	ARE	MODIFIED S MODULUS,	MOMENT OF INERTIA, 1 In.4	Ę	AR	MODIFIED S MODULUS,	MOMENT OF INERTIA, I In-4
3%" V				24.0	162.0	600.0	7,776	54.0	472.5	3,598.0	114,81
				25.5	172.1	672.8	9,327	55.5	485.6	3,789.1	124,65
6.0	<u>18.8</u> 23.4	18.8 29.3	56 110	27.0 28.5	182.3 192.4	749.5	11,072	57.0 58.5	498.8 511.9	3,984.9	135,03
<u>7.5</u> 9.0	23.4	42.2	190	30.0	202.5	914.5	15,188	60.0	525.0	4,185.3	145,98
10.5	32.8	57.4	302	31.5	212.6	1,002.8	17,581			1,370.3	101,00
12.0	37.5	75.0	450	33.0	222.8	1,094.9	20,215	10¾″ W	THIDTH		<u></u>
13.5	42.2	93.7	641	34.5	232.9	1,190.8	23,098	15.0	161.3	393.3	- 3,09
15.0	46.9	114.3	879	36.0	243.0	1,290.5	26,244	16.5	177.4	470.8	4,0
16.5 18.0	51.6	136.9	1,170	37.5	253.1 263.3	1,393.9	29,663	18.0	193.5 209.6	554.9 645.5	5,2
19.5	60.9	187.6	1,931	40.5	203.5	1,612.0	37,367	21.0	225.8	742.5	8,2
21.0	65.6	215.8	2,412	42.0	283.5	1,726.6	41,674	22.5	241.9	845.8	10,2
22.5	70.3	245.9	2,966	43.5	293.6	1,845.0	46,301	24.0	258.0	955.5	12,3
24.0	75.0	277.8	3,600	45.0	303.8	1,967.0	51,258	25.5	274.1	1,071.4	14,8
5%" W	/IDTH			46.5	313.9	2,092.6	56,556	27.0	290.3 306.4	1,193.6	17,6
7.5	38.4	48.0	180	1		2,222.0	1 02,200	1 30.0	322.5	1,456.4	24,1
9.0	45.1	69.2	311	8¾″ W	/IDTH	1997 - 1997 -		31.5	338.6	1,597.0	28,0
10.5	53.8	94.2	494	12.0	105.0	210.0	1,260	33.0	354.8	1,743.7	32,1
12.0	61.5	123.0	738	13.5	118.1	262.3	1,794	34.5	370.9	1,896.4	36,7
13.5	69.2	153.6	1,051	15.0	131.3	320.1	2,461	36.0	403.1	2,055.2	41,7
15.0 16.5	76.9 84.6	224.5	1,441	18.0	157.5	451.7	4,252	39.0	419.3	2,390.6	53,1
18.0	92.3	264.6	2,491	19.5	170.6	525.4	5,407	40.5	435.4	2,567.3	59,5
19.5	99.9	307.7	3,167	21.0	183.8	604.4	6,753	42.0	451.5	2,749.8	56,3
21.0	107.6	354.0	3,955	22.5	196.9	688.5	8,306	43.5	467.6	2,938.3	73,7
22.5	115.3	403.2	4,865	24.0	210.0	777.7	10,080	45.0	483.8	3,132.6	81,6
24.0	123.0	455.5 510.8	5,904	25.5	223.1	872.1 971.5	12,091	46.5	516.0	3,538.7	99,0
25.5 27.0	130.7	569.0	8,406	28.5	230.3	1,076.0	16,880	49.5	532.1	3,750.5	108,6
28.5	146.1	630.2	9,887	30.0	262.5	1,185.5	19,688	51.0	548.3	3,968.0	
30.0	153.8	694.3	11,531	31.5	275.6	1,299.9	22,791	52.5	564.4	4,191.4	129,6
31.5	161.4	761.4	13,349	33.0	288.8	1,419.3	26,204	54.0	580.5	4,420.4	141,0
33.0	169.1	831.3	15,348	34.5	<u>301.9</u> 315.0	1,543.6	29,942	55.5	612.8	4,055.2	165,9
34.5 36.0	175.8	904.1 979.8	19,926	37.5	328.1	1,806.9	38,452	58.5	628.9	5,141.9	
		,,,,,	.,,,20	39.0	341.3	1,945.9	43,253	60.0	645.0	5,398.8	193,5
6%" W	/IDTH			40.5	354.4	2,089.6	48,439	61.5	661.1	5,651.4	
120	<u>81.0</u>	162.0	972	42.0	367.5	2,238.2	54,022	63.0	677.3	5,914.5	224,0
13.5	91.1	202.4	. 1,384	43.5	380.6	2,391.6	60,020	64.5	693.4	6,183.3 6,457.8	257,5
150	101.3	246.9 295.6	<u>1,898</u> 2,527	45.0	393.8 406.9	2,549.8	66,445 73,314	67.5	725.6	6,737.8	275,5
16.5 18.0	111.4	348.4	3,280	48.0	420.0	2,880.3	80,640	69.0	741.8	7,023.4	
19.5	121.5	405.3	4,171	49.5	433.1	3,052.7	88,439	70.5	757.9	7,314.6	
21.0	141.8	466.2	5,209	51.0	446.3	3,229.8	96,725	72.0	774.0	7,611.3	334,3

# **Glue Laminated Timber**

These members come in nominal widths of 3, 4, 6, 8, 10, 12, 14 and 16 inches. The depth can exceed 12 inches, so the size factor,  $C_F$  must be used. The formula is based on a uniformly loaded beam, simply supported with an l/d ratio of 21. With a single midspan load, multiply  $C_F$  by 1.078. With two loads at third points, multiply  $C_F$  by 0.968. (Note: the table on page 4 provides section modulus that include  $C_F$ ).

$$C_F = (12/d)^{\frac{1}{9}} \le 1.0$$

If a glulam is subject to lateral buckling, the slenderness factor is used, and the size factor is not.

Bending of a curved glulam causes radial stresses (like membrane pressures) in tension and compression which can be evaluated for an arc with a radius of R at the neutral axis from:

$$f_r = \frac{3M}{2Rbd}$$
 for constant rectangular cross section

$$f_r \leq F_R$$
 where  $F_R = \begin{cases} F_{C\perp} \\ \frac{1}{3}F_V \end{cases}$ 

# **ASD Beam Design Flow Chart**

