

## Wood Design

### Notation:

$a$	= name for width dimension	$F_c^*$	= intermediate compressive stress for column design dependent on load duration
$A$	= name for area	$F_{cE}$	= theoretical allowed buckling stress
$A_{req'd-adj}$	= area required at allowable stress when shear is adjusted to include self weight	$F_{c\perp}$	= tabular compression strength perpendicular to the grain
$b$	= width of a rectangle = name for height dimension	$F_p$	= tabular bearing strength parallel to the grain = allowable bearing stress
$c_1$	= coefficient for shear stress for a rectangular bar in torsion	$F_R$	= allowable radial stress
$C_C$	= curvature factor for laminated arches	$F_t$	= tabular tensile strength
$C_D$	= load duration factor	$F_u$	= ultimate strength
$C_{fu}$	= flat use factor for other than decks	$F_v$	= tabular bending strength = allowable shear stress
$C_F$	= size factor	$h$	= height of a rectangle
$C_H$	= shear stress factor	$I$	= moment of inertia with respect to neutral axis bending
$C_i$	= incising factor	$I_{trial}$	= moment of inertia of trial section
$C_L$	= beam stability factor	$I_{req'd}$	= moment of inertia required at limiting deflection
$C_M$	= wet service factor	$I_y$	= moment of inertia with respect to an y-axis
$C_p$	= column stability factor for wood design	$J$	= polar moment of inertia
$C_r$	= repetitive member factor for wood design	$K_{cE}$	= material factor for wood column design
$C_V$	= volume factor for glue laminated timber design	$L_e$	= effective length that can buckle for column design, as is $\ell_e$
$C_t$	= temperature factor for wood design	$L$	= name for length or span length
$d$	= name for depth	$LL$	= shorthand for live load
$d_{min}$	= dimension of timber critical for buckling	$LRFD$	= load and resistance factor design
$DL$	= shorthand for dead load	$M$	= internal bending moment
$E$	= modulus of elasticity	$M_{max}$	= maximum internal bending moment
$f$	= stress (strength is a stress limit)	$M_{max-adj}$	= maximum bending moment adjusted to include self weight
$f_b$	= bending stress	$P$	= name for axial force vector
$f_{from\ table}$	= tabular strength (from table)	$R$	= radius of curvature of a deformed beam = radius of curvature of a laminated arch
$f_p$	= bearing stress		= name for a reaction force
$f_r$	= radial stress for a glulam timber	$S$	= section modulus
$f_v$	= shear stress	$S_{req'd}$	= section modulus required at allowable stress
$f_{v-max}$	= maximum shear stress		
$F_b$	= tabular bending strength = allowable bending stress		
$F'_b$	= allowable bending stress (adjusted)		
$F_c$	= tabular compression strength parallel to the grain		
$F'_c$	= allowable compressive stress (adjusted)		

$S_{req'd-adj}$  = section modulus required at allowable stress when moment is adjusted to include self weight  
 $T$  = torque (axial moment)  
 $V$  = internal shear force  
 $V_{max}$  = maximum internal shear force  
 $V_{max-adj}$  = maximum internal shear force adjusted to include self weight  
 $w$  = name for distributed load

$w_{self\ wt}$  = name for distributed load from self weight of member  
 $\Delta_{allowable}$  = allowable beam deflection  
 $\Delta_{limit}$  = allowable beam deflection limit  
 $\Delta_{max}$  = maximum beam deflection  
 $\kappa$  = slenderness ratio limit for long columns  
 $\gamma$  = density or unit weight  
 $\rho$  = radial distance

### Wood or Timber Design

Structural design standards for wood are established by the *National Design Specification (NDS)* published by the National Forest Products Association. There is a combined specification (from 2005) for **Allowable** Stress Design and limit state design (LRFD).

Tabulated wood strength values are used as the base allowable strength (ASD) and modified by appropriate adjustment factors:

$$f = C_D C_M C_F \dots \times f_{from\ table}$$

#### Adjustment Factors

- $C_D$  load duration factor
- $C_M$  wet service factor (1.0 dry < 16% moisture content)
- $C_t$  temperature factor (at high temperatures strength decreases)
- $C_L$  beam stability factor (for beams without full lateral support)
- $C_F$  size factor for visually graded sawn lumber and round timber > 12" depth

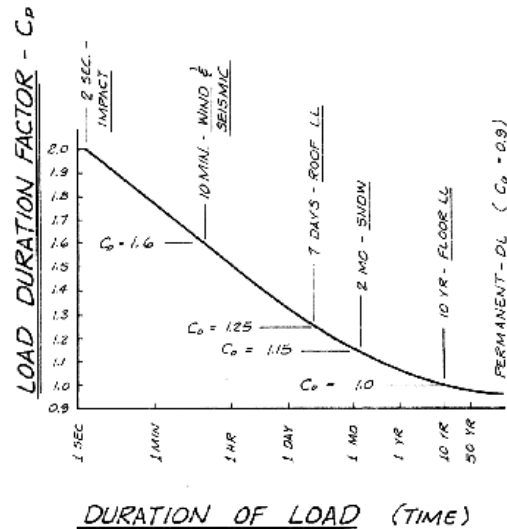
$$C_F = (12 / d)^{1/9} \leq 1.0$$

- $C_V$  volume factor for glued laminated timber (similar to  $C_F$ )
- $C_{fu}$  flat use factor (excluding decking)
- $C_r$  repetitive member factor (1.15 for three or more parallel members of Dimension lumber spaced not more than 24 in. on center, connected together by a load-distributing element such as roof, floor, or wall sheathing)
- $C_c$  curvature factor for glued laminated timber (1.0 straight & cambered)

$t/R \leq 1/100$  for hardwoods & southern pine or  $1/125$  other softwoods

$$C_c = 1 - 2000(t / R)^2$$

- $C_i$  incising factor (0.85 incised sawn lumber, 1 for sawn lumber not incised and glulam)
- $C_H$  shear stress factor (amount of splitting)
- $C_P$  column stability factor (1.0 for fully supported columns)



Design Values

- $F_b$ : bending stress
- $F_t$ : tensile stress
- $F_v$ : horizontal shear stress
- $F_{c\perp}$ : compression stress (perpendicular to grain)
- $F_c$ : compression stress (parallel to grain)
- $E$ : modulus of elasticity
- $F_p$ : bearing stress (parallel to grain)

Wood is significantly weakest in shear and strongest along the direction of the grain (tension and compression).

Load Combinations and Deflection

The critical load combination is determined by the largest of either:

$$\frac{\text{dead load}}{0.9} \text{ or } \frac{(\text{dead load} + \text{any combination of live load})}{C_D}$$

The deflection limits may be increased for less stiffness with total load:  $LL + 0.5(DL)$

**Criteria for Beam Design**

Allowable normal stress or normal stress from LRFD should not be exceeded:

Knowing  $M$  and  $F_b$ , the minimum section modulus fitting the limit is:  $S_{req'd} \geq \frac{M}{F_b}$

Besides strength, we also need to be concerned about *serviceability*. This involves things like limiting deflections & cracking, controlling noise and vibrations, preventing excessive settlements of foundations and durability. When we know about a beam section and its material, we can determine beam deformations.

Determining Maximum Bending Moment

Drawing  $V$  and  $M$  diagrams will show us the maximum values for design. Computer applications are very helpful.

Determining Maximum Bending Stress

For a prismatic member (constant cross section), the maximum normal stress will occur at the maximum moment.

For a *non-prismatic* member, the stress varies with the cross section AND the moment.

## Deflections

Elastic curve equations can be found in handbooks, textbooks, design manuals, etc...Computer programs can be used as well.

Elastic curve equations can be superpositioned **ONLY** if the stresses are in the elastic range. *The deflected shape is roughly the same shape flipped as the bending moment diagram but is constrained by supports and geometry.*

## Allowable Deflection Limits

All building codes and design codes limit deflection for beam types and damage that could happen based on service condition and severity.

Use	LL only	DL+LL
Roof beams:		
Industrial	L/180	L/120
Commercial		
plaster ceiling	L/240	L/180
no plaster	L/360	L/240
Floor beams:		
Ordinary Usage	L/360	L/240
Roof or floor (damageable elements)		L/480

## Lateral Buckling

With compression stresses in the top of a beam, a sudden “popping” or buckling can happen even at low stresses. In order to prevent it, we need to brace it along the top, or laterally brace it, or provide a bigger  $I_y$ .

## Beam Loads & Load Tracing

In order to determine the loads on a beam (or girder, joist, column, frame, foundation...) we can start at the top of a structure and determine the *tributary area* that a load acts over and the beam needs to support. Loads come from material weights, people, and the environment. This area is assumed to be from half the distance to the next beam over to halfway to the next beam.

The reactions must be supported by the next lower structural element *ad infinitum*, to the ground.

## Design Procedure

The intent is to find the most light weight member satisfying the section modulus size.

1. Know  $F_{all}$  for the material or  $F_U$  for LRFD.
2. Draw V & M, finding  $M_{max}$ .

3. Calculate  $S_{req'd}$ . This step is equivalent to determining  $f_b = \frac{M_{max}}{S} \leq F'_b$

4. For rectangular beams  $S = \frac{bh^2}{6}$

- For timber: use the section charts to find  $S$  that will work *and remember that the beam self weight will increase  $S_{req'd}$ .*

\*\*\*\*Determine the “updated”  $V_{max}$  and  $M_{max}$  including the beam self weight, and verify that the updated  $S_{req'd}$  has been met. \*\*\*\*

5. Consider lateral stability.

6. Evaluate horizontal shear stresses using  $V_{max}$  to determine if  $f_v \leq F'_v$

For rectangular beams  $f_{v-max} = \frac{3V}{2A} = 1.5 \frac{V}{A}$

7. Provide adequate bearing area at supports:  $f_p = \frac{P}{A} \leq F'_p$

8. Evaluate shear due to torsion  $f_v = \frac{T\rho}{J}$  or  $\frac{T}{c_1ab^2} \leq F'_v$

(circular section or rectangular)

9. Evaluate the deflection to determine if  $\Delta_{maxLL} \leq \Delta_{LL-allowed}$  and/or  $\Delta_{maxTotal} \leq \Delta_{Total-allowed}$

\*\*\*\* note: when  $\Delta_{calculated} > \Delta_{limit}$ ,  $I_{required}$  can be found with:  
and  $S_{req'd}$  will be satisfied for similar self weight \*\*\*\*  $I_{req'd} \geq \frac{\Delta_{toobig}}{\Delta_{limit}} I_{trial}$

FOR ANY EVALUATION:

Redesign (with a new section) at any point that a stress or serviceability criteria is NOT satisfied and re-evaluate each condition until it is satisfactory.

### Load Tables for Uniformly Loaded Joists & Rafters

Tables exist for the common loading situation for joists and rafters – that of uniformly distributed load. The tables either provide the safe distributed load based on bending and deflection limits, they give the allowable span for specific live and dead loads. If the load is *not uniform*, an *equivalent distributed load* can be calculated from the maximum moment equation.

### Decking

Flat panels or planks that span several joists or evenly spaced support behave as continuous beams. Design tables consider a “1 unit” wide strip across the supports and determine maximum bending moment and deflections in order to provide allowable loads depending on the depth of the material.

The other structural use of decking is to construct what is called a *diaphragm*, which is a horizontal or vertical (if the panels are used in a shear wall) unit tying the sheathing to the joists or studs that resists forces parallel to the surface of the diaphragm.

## Column Design

If we know the loads, we can select a section that is adequate for strength & buckling.

If we know the length, we can find the limiting load satisfying strength & buckling.

Any slenderness ratio,  $L_e/d \leq 50$ :

$$f_c = \frac{P}{A} \leq F'_c \qquad F'_c = F_c (C_D)(C_M)(C_t)(C_F)(C_p)$$

The allowable stress equation uses factors to replicate the combination crushing-buckling curve:

where:

$F'_c$  = allowable compressive stress parallel to the grain

$F_c$  = compressive strength parallel to the grain

$C_D$  = load duration factor

$C_M$  = wet service factor (1.0 for dry)

$C_t$  = temperature factor

$C_F$  = size factor

$C_p$  = column stability factor off chart  
or equation:

$$C_p = \frac{1 + (F_{cE} / F_c^*)}{2c} - \sqrt{\left[ \frac{1 + F_{cE} / F_c^*}{2c} \right]^2 - \frac{F_{cE} / F_c^*}{c}}$$

For preliminary column design:

$$F'_c = F_c^* C_p = (F_c C_D) C_p$$

## Procedure for Analysis

1. Calculate  $L_e/d_{\min}$  ( $KL/d$  for each axis and chose largest)
2. Obtain  $F'_c$

$$\text{compute } F_{cE} = \frac{K_{cE} E}{(l_e/d)^2} \quad \text{with } K_{cE} = 0.3 \text{ for sawn, } = 0.418 \text{ for glu-lam}$$

3. Compute  $F_c^* \cong F_c C_D$  with  $C_D = 1$ , normal,  $C_D = 1.25$  for 7 day roof, etc....
4. Calculate  $F_{cE} / F_c^*$  and get  $C_p$  from table or calculation
5. Calculate  $F'_c = F_c^* C_p$
6. Compute  $P_{\text{allowable}} = F'_c \cdot A$  or alternatively compute  $f_{\text{actual}} = P/A$
7. Is the design satisfactory?

Is  $P \leq P_{\text{allowable}}$ ?  $\Rightarrow$  yes, it is; no, it is no good

or Is  $f_{\text{actual}} \leq F'_c$ ?  $\Rightarrow$  yes, it is; no, it is no good

### Procedure for Design

1. Guess a size by picking a section
2. Calculate  $L_e/d_{\text{min}}$  (KL/d for each axis and choose largest)
3. Obtain  $F'_c$

$$\text{compute } F_{cE} = \frac{K_{cE}E}{(l_e/d)^2} \text{ with } K_{cE} = 0.3 \text{ for sawn, } = 0.418 \text{ for glu-lam}$$

4. Compute  $F_c^* \cong F_c C_D$  with  $C_D = 1$ , normal,  $C_D = 1.25$  for 7 day roof...
5. Calculate  $F_{cE}/F_c^*$  and get  $C_p$  from table or calculation
6. Calculate  $F'_c = F_c^* C_p$
7. Compute  $P_{\text{allowable}} = F'_c \cdot A$  or alternatively compute  $f_{\text{actual}} = P/A$
8. Is the design satisfactory?

Is  $P \leq P_{\text{allowable}}$ ?  $\Rightarrow$  yes, it is; no, pick a bigger section and go back to step 2.

or Is  $f_{\text{actual}} \leq F'_c$ ?  $\Rightarrow$  yes, it is; no, pick a bigger section and go back to step 2.

### Trusses

Timber trusses are commonly manufactured with continuous top or bottom chords, but the members are still design as compression and tension members (without the effect of bending.)

### Stud Walls

Stud wall construction is often used in *light frame construction* together with joist and rafters. Studs are typically 2-in. nominal thickness and must be braced in the weak axis. Most wall coverings provide this function. Stud spacing is determined by the width of the panel material, and is usually 16 in. The lumber grade can be relatively low. The walls must be designed for a combination of wind load and bending, which means beam-column analysis.

### Columns with Bending (Beam-Columns)

The modification factors are included in the form:

$$\left[ \frac{f_c}{F'_c} \right]^2 + \frac{f_{bx}}{F'_{bx} \left[ 1 - \frac{f_c}{F_{cEx}} \right]} \leq 1.0$$

where:

$$1 - \frac{f_c}{F_{cEx}} = \text{magnification factor accounting for P-}\Delta$$

$F'_{bx}$  = allowable bending stress

$f_{bx}$  = working stress from bending about x-x axis

In order to *design* an adequate section for allowable stress, we have to start somewhere:

1. Make assumptions about the limiting stress from:
  - buckling
  - axial stress
  - combined stress
2. See if we can find values for  $\underline{r}$  or  $\underline{A}$  or  $\underline{S} (=I/c_{\max})$
3. Pick a trial section based on if we think  $r$  or  $A$  is going to govern the section size.
4. Analyze the stresses and compare to allowable using the allowable stress method or interaction formula for eccentric columns.
5. Did the section pass the stress test?
  - If not, do you *increase*  $r$  or  $A$  or  $S$ ?
  - If so, is the difference really big so that you could *decrease*  $r$  or  $A$  or  $S$  to make it more efficient (economical)?
6. Change the section choice and go back to step 4. Repeat until the section meets the stress criteria.

### Glue Laminated Timber

These members come in nominal widths of 3, 4, 6, 8, 10, 12, 14 and 16 inches. The depth can exceed 12 inches, so the size factor,  $C_F$  must be used. The formula is based on a uniformly loaded beam, simply supported with an  $l/d$  ratio of 21. With a single midspan load, multiply  $C_F$  by 1.078. With two loads at third points, multiply  $C_F$  by 0.968. (Note: The Section Properties/Standard Sizes table provides section modulus that include  $C_F$ ).

$$C_F = (12/d)^{1/9} \leq 1.0$$

If a glulam is subject to lateral buckling, the slenderness factor is used, and the size factor is not.

### Laminated Arches

The radius of curvature,  $R$ , is limited because of residual bending stresses between lams of thickness  $t$  to  $100t$  for Southern pine and hardwoods and  $250t$  for softwoods.

The allowable bending stress for combined stresses is  $F'_b = F_b (C_F C_C)$

Bending of a curved glulam causes radial stresses (like membrane pressures) in tension and compression which can be evaluated for an arc with a radius of  $R$  at the neutral axis from:

$$f_r = \frac{3M}{2Rbd} \quad \text{for constant rectangular cross section}$$

$$f_r \leq F_R \quad \text{where } F_R = \begin{cases} F_{C\perp} \\ \frac{1}{3} F_V \end{cases}$$



Table 9.3 Column stability factor  $C_p$ .

Statics and Strength of Materials for Architecture and Building Construction, 2nd ed., Onouye & Kane

Column Stability Factor  $C_p$

$C_p$			$F_c' = C_p \cdot F_c$			$F_{CE} = \frac{30 E}{(L/d)^2}$ for sawn posts			$F_{CE} = \frac{418 E}{(L/d)^2}$ for Glu-Lam posts		
$\frac{F_{CE}}{F_c}$	Sawn $C_p$	Glu-Lam $C_p$	$\frac{F_{CE}}{F_c}$	Sawn $C_p$	Glu-Lam $C_p$	$\frac{F_{CE}}{F_c}$	Sawn $C_p$	Glu-Lam $C_p$	$\frac{F_{CE}}{F_c}$	Sawn $C_p$	Glu-Lam $C_p$
0.00	0.000	0.000	0.60	0.500	0.538	1.20	0.750	0.822	2.40	0.894	0.940
0.01	0.010	0.010	0.61	0.506	0.545	1.22	0.755	0.826	2.45	0.897	0.941
0.02	0.020	0.020	0.62	0.512	0.552	1.24	0.760	0.831	2.50	0.899	0.943
0.03	0.030	0.030	0.63	0.518	0.559	1.26	0.764	0.836	2.55	0.901	0.944
0.04	0.040	0.040	0.64	0.524	0.566	1.28	0.769	0.840	2.60	0.904	0.946
0.05	0.049	0.050	0.65	0.530	0.573	1.30	0.773	0.844	2.65	0.906	0.947
0.06	0.059	0.060	0.66	0.536	0.580	1.32	0.777	0.848	2.70	0.908	0.949
0.07	0.069	0.069	0.67	0.542	0.587	1.34	0.781	0.852	2.75	0.910	0.950
0.08	0.079	0.079	0.68	0.548	0.593	1.36	0.785	0.855	2.80	0.912	0.951
0.09	0.088	0.089	0.69	0.553	0.600	1.38	0.789	0.859	2.85	0.914	0.952
0.10	0.098	0.099	0.70	0.559	0.607	1.40	0.793	0.862	2.90	0.916	0.953
0.11	0.107	0.109	0.71	0.564	0.613	1.42	0.796	0.865	2.95	0.917	0.954
0.12	0.117	0.118	0.72	0.569	0.619	1.44	0.800	0.868	3.00	0.919	0.955
0.13	0.126	0.128	0.73	0.575	0.626	1.46	0.803	0.871	3.05	0.920	0.956
0.14	0.136	0.138	0.74	0.580	0.632	1.48	0.807	0.874	3.10	0.922	0.957
0.15	0.145	0.147	0.75	0.585	0.638	1.50	0.810	0.877	3.15	0.923	0.958
0.16	0.154	0.157	0.76	0.590	0.644	1.52	0.813	0.879	3.20	0.925	0.959
0.17	0.164	0.167	0.77	0.595	0.650	1.54	0.816	0.882	3.25	0.926	0.960
0.18	0.173	0.176	0.78	0.600	0.655	1.56	0.819	0.884	3.30	0.927	0.961
0.19	0.182	0.186	0.79	0.605	0.661	1.58	0.822	0.887	3.35	0.929	0.961
0.20	0.191	0.195	0.80	0.610	0.667	1.60	0.825	0.889	3.40	0.930	0.962
0.21	0.200	0.205	0.81	0.614	0.672	1.62	0.827	0.891	3.45	0.931	0.963
0.22	0.209	0.214	0.82	0.619	0.678	1.64	0.830	0.893	3.50	0.932	0.963
0.23	0.218	0.224	0.83	0.623	0.683	1.66	0.832	0.895	3.55	0.933	0.964
0.24	0.227	0.233	0.84	0.628	0.688	1.68	0.835	0.897	3.60	0.934	0.965
0.25	0.235	0.242	0.85	0.632	0.693	1.70	0.837	0.899	3.65	0.936	0.965
0.26	0.244	0.252	0.86	0.637	0.698	1.72	0.840	0.901	3.70	0.937	0.966
0.27	0.253	0.261	0.87	0.641	0.703	1.74	0.842	0.903	3.75	0.938	0.966
0.28	0.261	0.270	0.88	0.645	0.708	1.76	0.844	0.904	3.80	0.938	0.967
0.29	0.270	0.279	0.89	0.649	0.713	1.78	0.846	0.906	3.85	0.939	0.968
0.30	0.278	0.288	0.90	0.653	0.718	1.80	0.849	0.908	3.90	0.940	0.968
0.31	0.287	0.297	0.91	0.658	0.722	1.82	0.851	0.909	3.95	0.941	0.969
0.32	0.295	0.306	0.92	0.661	0.727	1.84	0.853	0.911	4.00	0.942	0.969
0.33	0.304	0.315	0.93	0.665	0.731	1.86	0.855	0.912	4.05	0.943	0.969
0.34	0.312	0.324	0.94	0.669	0.735	1.88	0.857	0.914	4.10	0.944	0.970
0.35	0.320	0.333	0.95	0.673	0.740	1.90	0.858	0.915	4.15	0.944	0.970
0.36	0.328	0.342	0.96	0.677	0.744	1.92	0.860	0.916	4.20	0.945	0.971
0.37	0.336	0.351	0.97	0.680	0.748	1.94	0.862	0.918	4.25	0.946	0.971
0.38	0.344	0.360	0.98	0.684	0.752	1.96	0.864	0.919	4.30	0.947	0.972
0.39	0.352	0.368	0.99	0.688	0.756	1.98	0.866	0.920	4.35	0.947	0.972
0.40	0.360	0.377	1.00	0.691	0.760	2.00	0.867	0.921	4.40	0.948	0.972
0.41	0.367	0.386	1.01	0.694	0.764	2.02	0.869	0.922	4.45	0.949	0.973
0.42	0.375	0.394	1.02	0.698	0.767	2.04	0.870	0.924	4.50	0.949	0.973
0.43	0.383	0.403	1.03	0.701	0.771	2.06	0.872	0.925	4.55	0.950	0.974
0.44	0.390	0.411	1.04	0.704	0.774	2.08	0.874	0.926	4.60	0.950	0.974
0.45	0.398	0.420	1.05	0.708	0.778	2.10	0.875	0.927	4.65	0.951	0.974
0.46	0.405	0.428	1.06	0.711	0.781	2.12	0.876	0.928	4.70	0.952	0.975
0.47	0.412	0.436	1.07	0.714	0.784	2.14	0.878	0.929	4.75	0.952	0.975
0.48	0.419	0.444	1.08	0.717	0.788	2.16	0.879	0.930	4.80	0.953	0.975
0.49	0.427	0.453	1.09	0.720	0.791	2.18	0.881	0.931	4.85	0.953	0.975
0.50	0.434	0.461	1.10	0.723	0.794	2.20	0.882	0.932	4.90	0.954	0.976
0.51	0.441	0.469	1.11	0.726	0.797	2.22	0.883	0.932	5.00	0.955	0.976
0.52	0.448	0.477	1.12	0.729	0.800	2.24	0.885	0.933	6.00	0.963	0.981
0.53	0.454	0.484	1.13	0.731	0.803	2.26	0.886	0.934	8.00	0.973	0.986
0.54	0.461	0.492	1.14	0.734	0.806	2.28	0.887	0.935	10.00	0.979	0.989
0.55	0.468	0.500	1.15	0.737	0.809	2.30	0.888	0.936	20.00	0.990	0.995
0.56	0.474	0.508	1.16	0.740	0.811	2.32	0.889	0.937	40.00	0.995	0.997
0.57	0.481	0.515	1.17	0.742	0.814	2.34	0.891	0.937	60.00	0.997	0.998
0.58	0.487	0.523	1.18	0.745	0.817	2.36	0.892	0.938	100.00	0.998	0.999
0.59	0.494	0.530	1.19	0.747	0.819	2.38	0.893	0.939	200.00	0.999	0.999

Table developed and permission for use granted by Professor Ed Lebert, Dept. of Architecture, University of Washington.

**SECTION PROPERTIES / STANDARD SIZES** To the extent that other

considerations will permit, the finished sizes of structural glued laminated timber as given in Table B constitute normal industry practice. Industry standards do, however, permit the use of any depth or width of glued laminated timber. Dimension lumber of 1½ in. net thickness is normally used for laminating straight members. The modified section modulus includes size factor ( $C_r$ ), and no further reduction of bending stress for size is needed.

DEPTH, d in.	AREA, A in. <sup>2</sup>	MODIFIED SECTION MODULUS, $SC_f$ in. <sup>3</sup>	MOMENT OF INERTIA, I in. <sup>4</sup>	DEPTH, d in.	AREA, A in. <sup>2</sup>	MODIFIED SECTION MODULUS, $SC_f$ in. <sup>3</sup>	MOMENT OF INERTIA, I in. <sup>4</sup>	DEPTH, d in.	AREA, A in. <sup>2</sup>	MODIFIED SECTION MODULUS, $SC_f$ in. <sup>3</sup>	MOMENT OF INERTIA, I in. <sup>4</sup>
<b>3¼" WIDTH</b>				24.0	162.0	600.0	7,776	54.0	472.5	3,598.0	114,818
6.0	18.8	18.8	56	25.5	172.1	672.8	9,327	55.5	485.6	3,789.1	124,654
7.5	23.4	29.3	110	27.0	182.3	749.5	11,072	57.0	498.8	3,984.9	135,037
9.0	28.1	42.2	190	28.5	192.4	830.0	13,021	58.5	511.9	4,185.3	145,980
10.5	32.8	57.4	302	30.0	202.5	914.5	15,188	60.0	525.0	4,390.3	157,500
12.0	37.5	75.0	450	31.5	212.6	1,002.8	17,581	<b>10¼" WIDTH</b>			
13.5	42.2	93.7	641	33.0	222.8	1,094.9	20,215	15.0	161.3	393.3	3,023
15.0	46.9	114.3	879	34.5	232.9	1,190.8	23,098	16.5	177.4	470.8	4,024
16.5	51.6	136.9	1,170	36.0	243.0	1,290.5	26,244	18.0	193.5	554.9	5,224
18.0	56.3	161.3	1,519	37.5	253.1	1,393.9	29,663	19.5	209.6	645.5	6,642
19.5	60.9	187.6	1,931	39.0	263.3	1,501.1	33,367	21.0	225.8	742.5	8,296
21.0	65.6	215.8	2,412	40.5	273.4	1,612.0	37,367	22.5	241.9	845.8	10,204
22.5	70.3	245.9	2,966	42.0	283.5	1,726.6	41,674	24.0	258.0	955.5	12,384
24.0	75.0	277.8	3,600	43.5	293.6	1,845.0	46,301	25.5	274.1	1,071.4	14,854
<b>5¼" WIDTH</b>				45.0	303.8	1,967.0	51,258	27.0	290.3	1,193.6	17,633
7.5	38.4	48.0	180	46.5	313.9	2,092.6	56,556	28.5	306.4	1,321.9	20,738
9.0	46.1	69.2	311	48.0	324.0	2,222.0	62,208	30.0	322.5	1,456.4	24,188
10.5	53.8	94.2	494	<b>8¼" WIDTH</b>				31.5	338.6	1,597.0	28,000
12.0	61.5	123.0	738	12.0	105.0	210.0	1,260	33.0	354.8	1,743.7	32,194
13.5	69.2	153.6	1,051	13.5	118.1	262.3	1,794	34.5	370.9	1,896.4	36,786
15.0	76.9	187.5	1,441	15.0	131.3	320.1	2,461	36.0	387.0	2,055.2	41,796
16.5	84.6	224.5	1,919	16.5	144.4	383.2	3,276	37.5	403.1	2,219.9	47,241
18.0	92.3	264.6	2,491	18.0	157.5	451.7	4,252	39.0	419.3	2,390.6	53,140
19.5	99.9	307.7	3,167	19.5	170.6	525.4	5,407	40.5	435.4	2,567.3	59,510
21.0	107.6	354.0	3,955	21.0	183.8	604.4	6,753	42.0	451.5	2,749.8	66,370
22.5	115.3	403.2	4,865	22.5	196.9	688.5	8,306	43.5	467.6	2,938.3	73,739
24.0	123.0	455.5	5,904	24.0	210.0	777.7	10,080	45.0	483.8	3,132.6	81,633
25.5	130.7	510.8	7,082	25.5	223.1	872.1	12,091	46.5	499.9	3,332.7	90,071
27.0	138.4	569.0	8,406	27.0	236.3	971.5	14,352	48.0	516.0	3,538.7	99,072
28.5	146.1	630.2	9,887	28.5	249.4	1,076.0	16,880	49.5	532.1	3,750.5	108,653
30.0	153.8	694.3	11,531	30.0	262.5	1,185.5	19,688	51.0	548.3	3,968.0	118,833
31.5	161.4	761.4	13,349	31.5	275.6	1,299.9	22,791	52.5	564.4	4,191.4	129,630
33.0	169.1	831.3	15,348	33.0	288.8	1,419.3	26,204	54.0	580.5	4,420.4	141,062
34.5	176.8	904.1	17,538	34.5	301.9	1,543.6	29,942	55.5	596.6	4,655.2	153,146
36.0	184.5	979.8	19,926	36.0	315.0	1,672.8	34,020	57.0	612.8	4,895.7	165,902
<b>6¼" WIDTH</b>				37.5	328.1	1,806.9	38,452	58.5	628.9	5,141.9	179,347
12.0	81.0	162.0	972	39.0	341.3	1,945.9	43,253	60.0	645.0	5,398.8	193,500
13.5	91.1	202.4	1,384	40.5	354.4	2,089.6	48,439	61.5	661.1	5,651.4	208,379
15.0	101.3	246.9	1,898	42.0	367.5	2,238.2	54,022	63.0	677.3	5,914.5	224,000
16.5	111.4	295.6	2,527	43.5	380.6	2,391.6	60,020	64.5	693.4	6,183.3	240,384
18.0	121.5	348.4	3,280	45.0	393.8	2,549.8	66,445	66.0	709.5	6,457.8	257,548
19.5	131.6	405.3	4,171	46.5	406.9	2,712.7	73,314	67.5	725.6	6,737.8	275,511
21.0	141.8	466.2	5,209	48.0	420.0	2,880.3	80,640	69.0	741.8	7,023.4	294,289
22.5	151.9	531.1	6,407	49.5	433.1	3,052.7	88,439	70.5	757.9	7,314.6	313,902
				51.0	446.3	3,229.8	96,725	72.0	774.0	7,611.3	334,368
				52.5	459.4	3,411.6	105,513	73.5	790.1	7,913.6	355,704



### ASD Beam Design Flow Chart

